FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL HARDWARE

NUMBER: 05-68A-2116 -X

SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL

REVISION: 2

03/19/90

PART DATA

PART NAME VENDOR NAME

PART NUMBER
VENDOR NUMBER

LRU

: PANEL A12

V070-730365

SRU

: SWITCH, TOGGLE

ME452-0102-7256

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

SWITCH, TOGGLE, 2P2P, LANDING GEAR ARM/DOWN RESET, (MOM)

REFERENCE DESIGNATORS:

36V73A12S13

QUANTITY OF LIKE ITEMS: 1

ONE

FUNCTION:

PROVIDES MANUAL MAINTAINED OFF AND MOMENTARY ON POWER TO THE LANDING GEAR ARM/DOWN RESET CIRCUIT (RESETS LATCHING RELAYS IN APPLICABLE CIRCUITS). ONE POLE OF SWITCH IS USED IN INPUT LINE WITH THE OTHER IN THE RETURN LINE. USE OF RESET CIRCUIT IS NORMALLY A GROUND FUNCTION.

FAILURE MODES EFFECTS ANALYSIS FMEA - CIL FAILURE MODE NUMBER: 05-6BA-2116-03 REVISION#: 06/28/99 SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL LRU: PANEL A12 **CRITICALITY OF THIS** ITEM NAME: SWITCH, TOGGLE FAILURE MODE: 1R2 **FAILURE MODE:** FAILS CLOSED (BOTH POLES) DO DE-ORBIT MISSION PHASE: 102 COLUMBIA VEHICLE/PAYLOAD/KIT EFFECTIVITY: 103 DISCOVERY 104 **ATLANTIS ENDEAVOUR** 105 CAUSE: PIECE PART STRUCTURAL FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK, PROCESSING ANOMALY CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO REDUNDANCY SCREEN A) PASS B) FAIL C) PASS PASS/FAIL RATIONALE: A) B) FAILS "B" SCREEN BECAUSE SWITCH FAILURE IS NOT FLIGHT DETECTABLE SINCE THE CIRCUIT BREAKER IS NORMALLY OPEN. C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

FIRST FAILURE - NO EFFECT

(B) INTERFACING SUBSYSTEM(S):

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FIRST FAILURE - NO EFFECT. SECOND FAILURE (CIRCUIT BREAKER 60 ON PANEL R15 FAILS CLOSED WHERE IT IS NORMALLY OPEN) - CAUSES LANDING GEAR ARM AND DOWN RELAYS TO BE PERMANENTLY RESET.

(C) MISSION:

FIRST FAILURE - NO EFFECT

(D) CREW, VEHICLE, AND ELEMENT(\$):

FIRST FAILURE - NO EFFECT. SECOND FAILURE (CIRCUIT BREAKER 60 ON PANEL R15 FAILS CLOSED WHERE IT IS NORMALLY OPEN) - CAUSES LANDING GEAR ARM AND DOWN RELAYS TO BE PERMANENTLY RESET. THIS PREVENTS EXTENSION OF LANDING GEARS AND CAUSES LOSS OF CREW/VEHICLE BECAUSE ARM AND DOWN SWITCHES CAN NOT SET THE ARM AND DOWN RELAYS REQUIRED FOR GEAR EXTENSION.

(E) FUNCTIONAL CRITICALITY EFFECTS:

-DISPOSITION RATIONALE-

(A) DESIGN:

REFER TO APPENDIX A, ITEM NO. 1 - TOGGLE SWITCH

(B) TEST:

REFER TO APPENDIX A, ITEM NO. 1 - TOGGLE SWITCH

GROUND TURNAROUND TEST

ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD.

(C) INSPECTION:

RÉFER TO APPENDIX A, ITEM NO. 1 - TOGGLE SWITCH

(D) FAILURE HISTORY:

CURRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATA BASE.

(E) OPERATIONAL USE:

NONE

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- APPROVALS -

EDITORIALLY APPROVED

TECHNICAL APPROVAL

: BNA

: VIA APPROVAL FORM

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